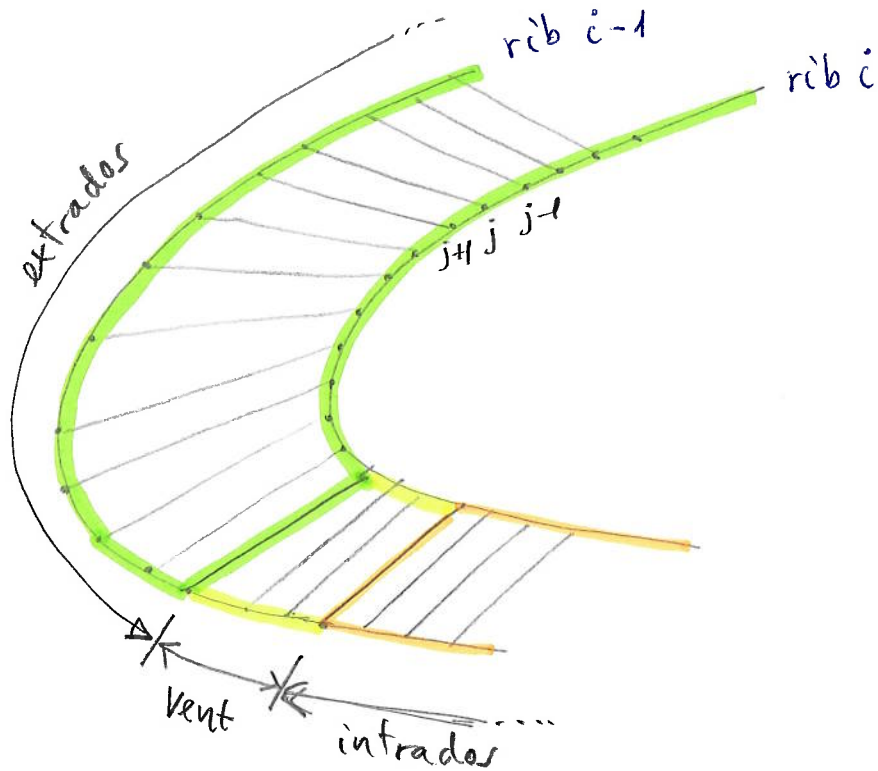


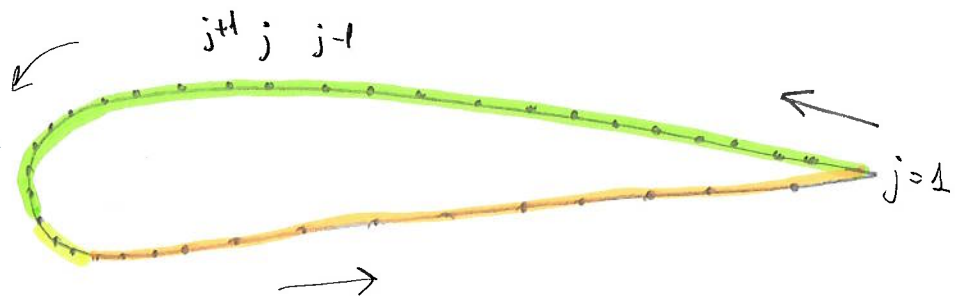
# 3D-SHAPING PROGRAMMING STRATEGY

For each panel between rib  $i-1$  and rib  $i$

Ribs  $0, 1, 2, \dots, i, \dots, n_{\text{maxribs}}$



Each  
airfoil  
have  
 $n$  points

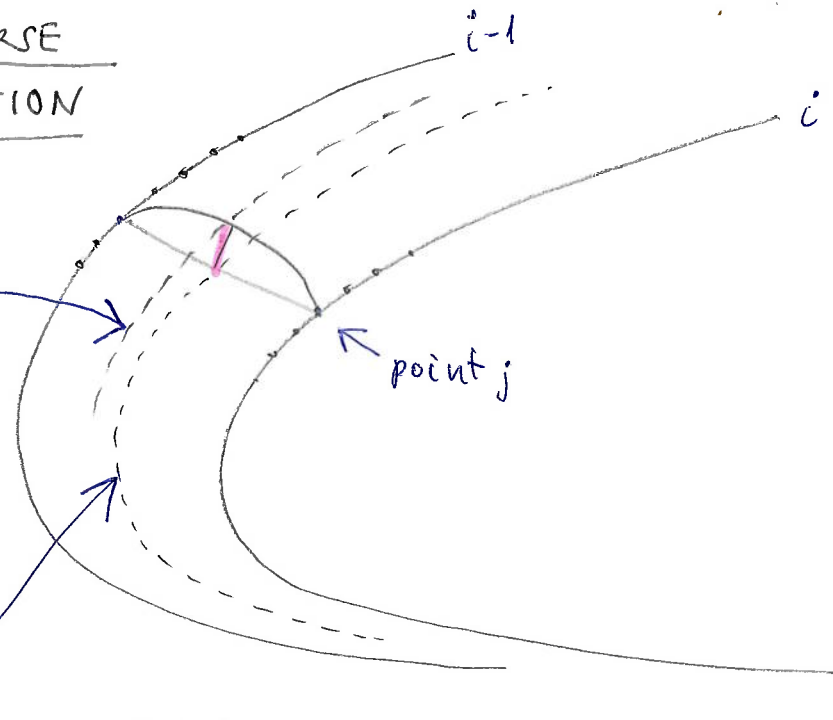


$j = 1, 2, 3, \dots, j, \dots, n_{\text{points}}$

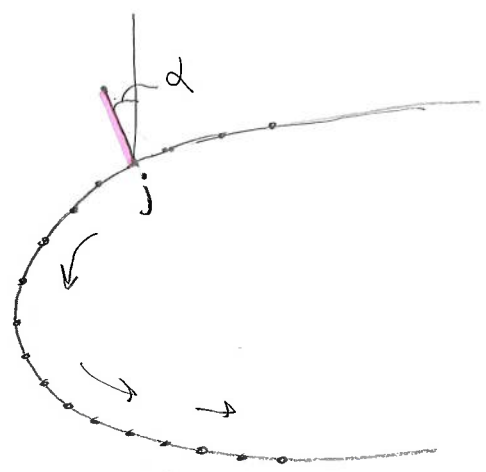
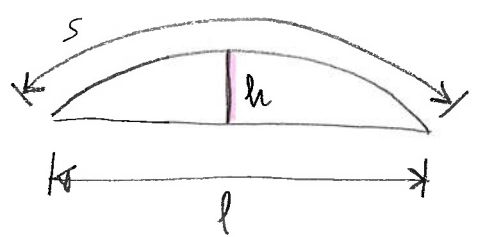
Consider:

TRANSVERSE  
OVALIZATION

- Ovalized airfoil between  $i-1$  and  $i$



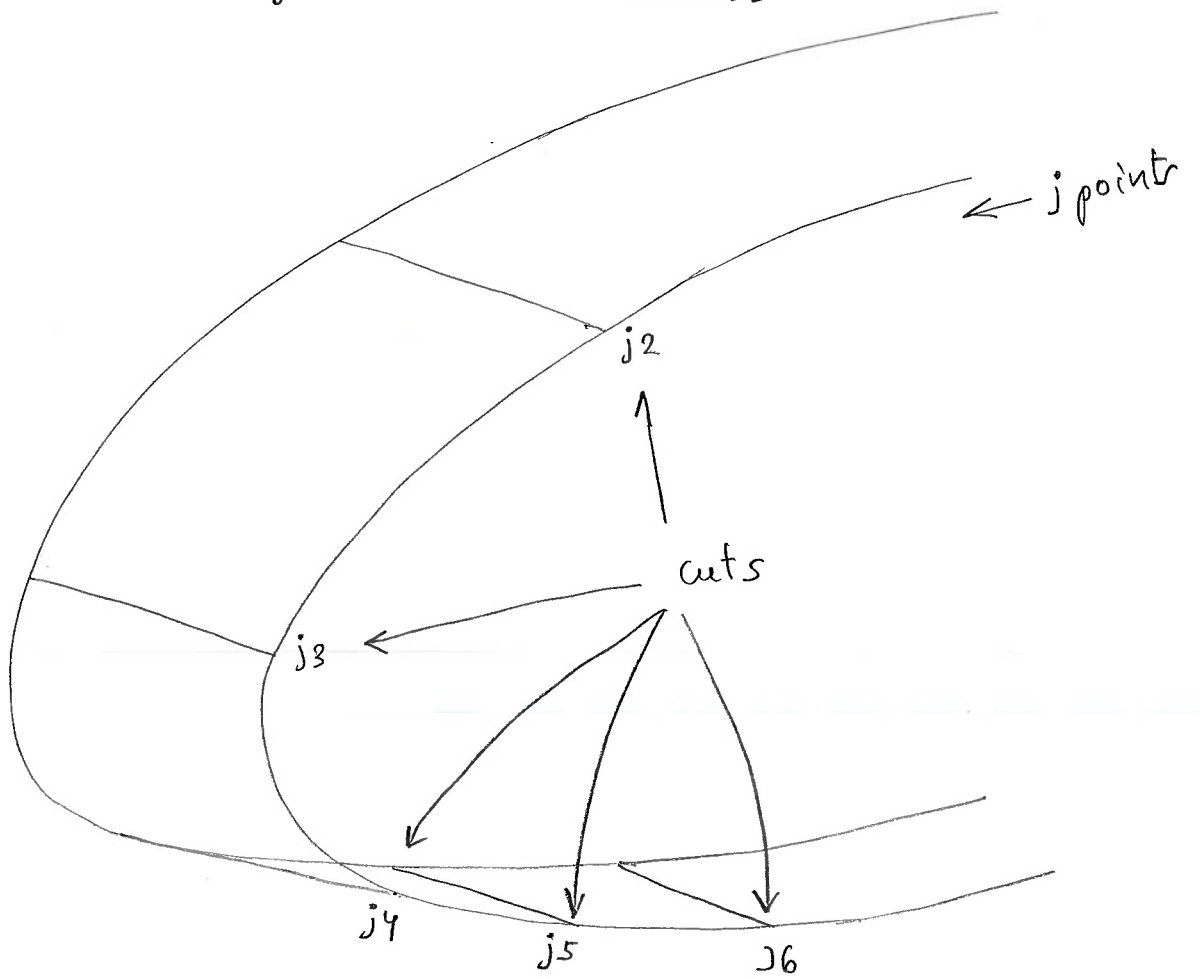
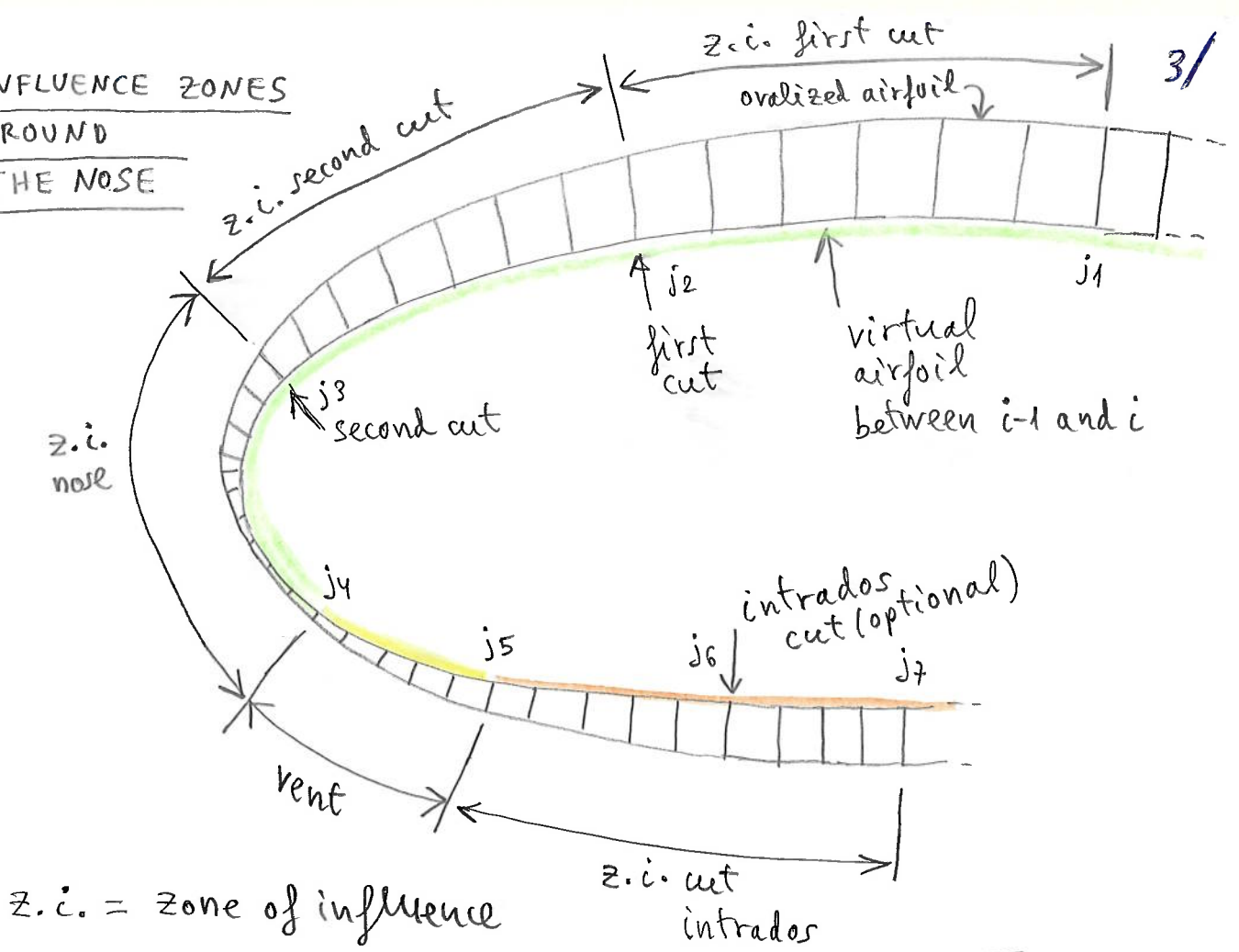
- Virtual airfoil between  $i-1$  and  $i$
- straight distance between corresponding points  $j$  in airfoil  $i-1$  and  $i$  ( $l$ )



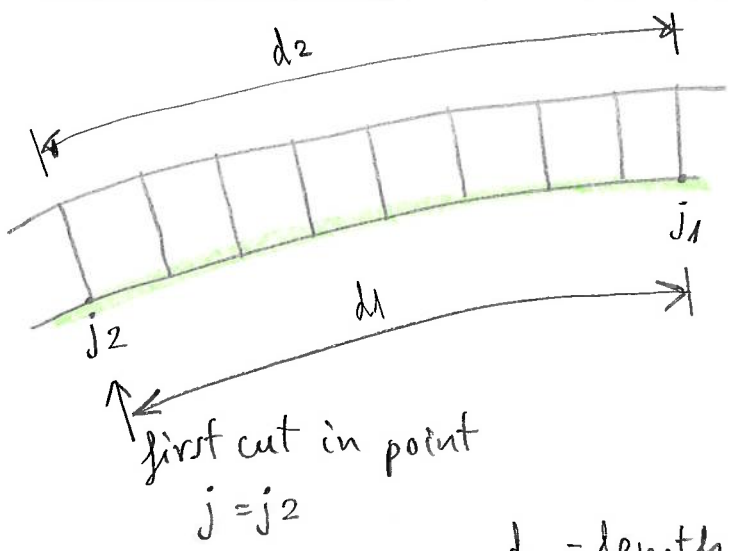
- Ovalized distance ( $s$ )
- Ovalization height ( $h$ )
- Sector plane angle ( $\alpha$ )

} transverse ovalization parameters

INFLUENCE ZONES  
AROUND  
THE NOSE



# Consider z.c. of first cut in detail



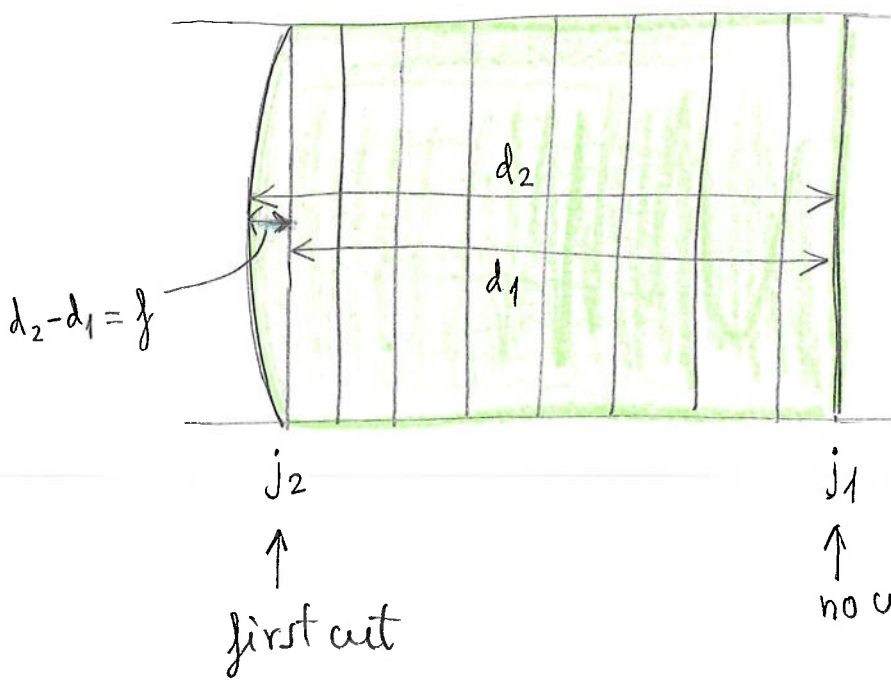
$$(\Delta_1 = d_2 - d_1)$$

$d_1$  = length between  $j_1$  and  $j_2$  along virtual airfoil

$d_2$  = length between  $j_1$  and  $j_2$  along ovalized airfoil

In most cases  $d_2 > d_1$  then...  
 if we want the panel have a uniform tension, we need to extend its center part

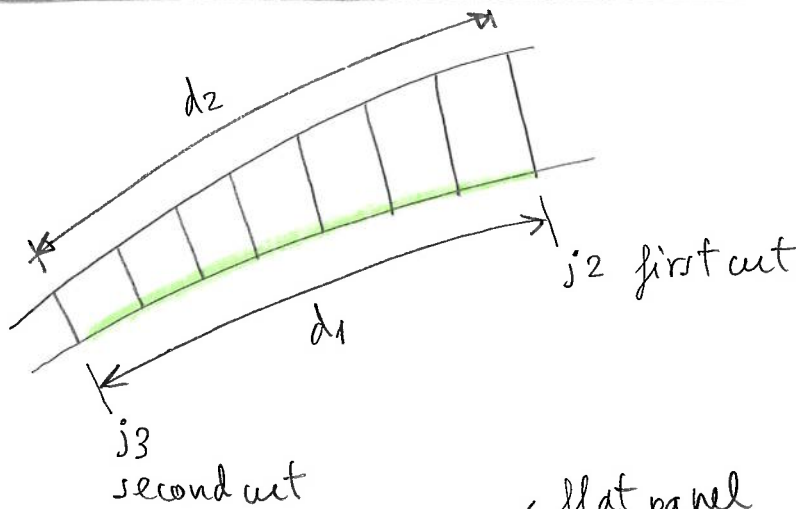
3D-shaping strategy



$j_2$   
↑  
first cut

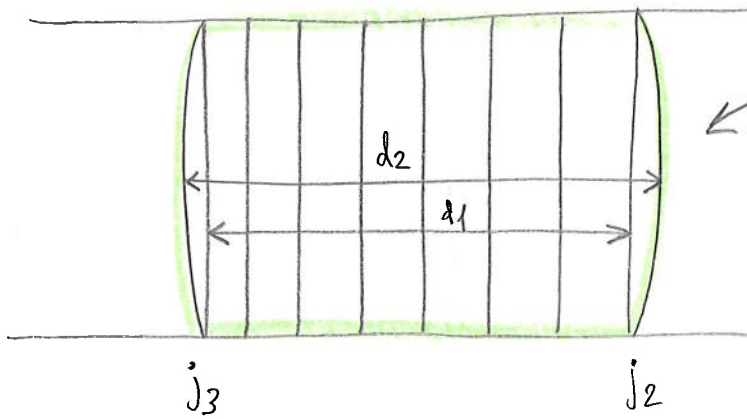
$j_1$   
↑  
no cut (extrados panel)

Consider z.i. of second cut in detail



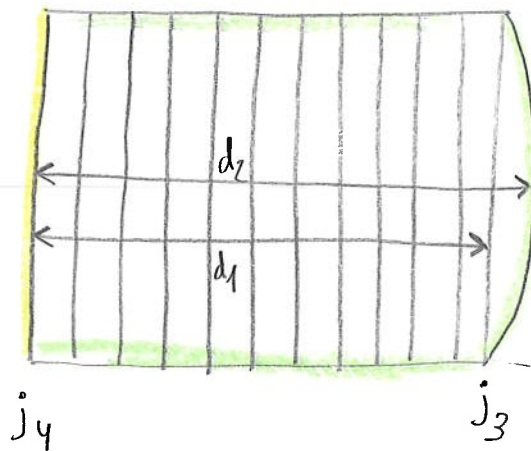
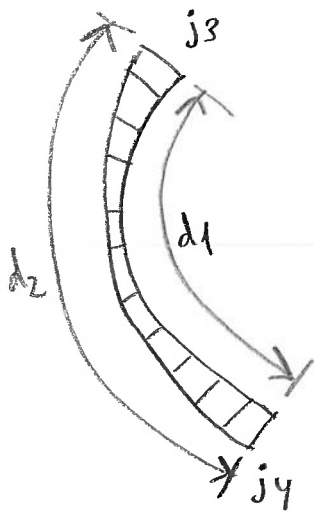
$d_2 > d_1$   
 $(\Delta_2 = d_2 - d_1)$

flat panel



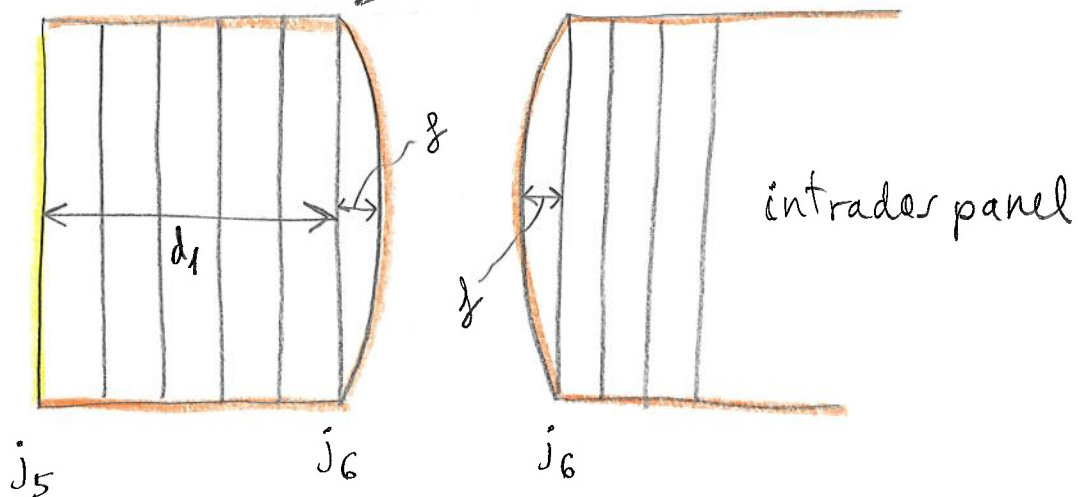
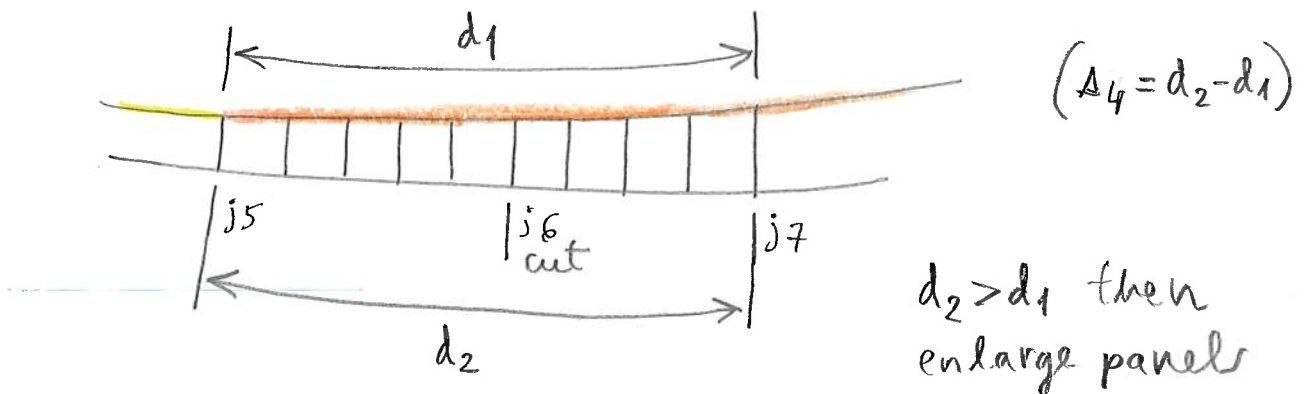
3D-shaping strategy:  
 We enlarge the panel in its central part to avoid stretching

Consider z.i. of nose in detail



$d_2 > d_1$   
 $(\Delta_3 = d_2 - d_1)$

Consider z.c. of intrados cut in detail



geometric compatibility:  $f_{\text{left}} = f_{\text{right}} = f$

then  $d_2 = d_1 + 2f$

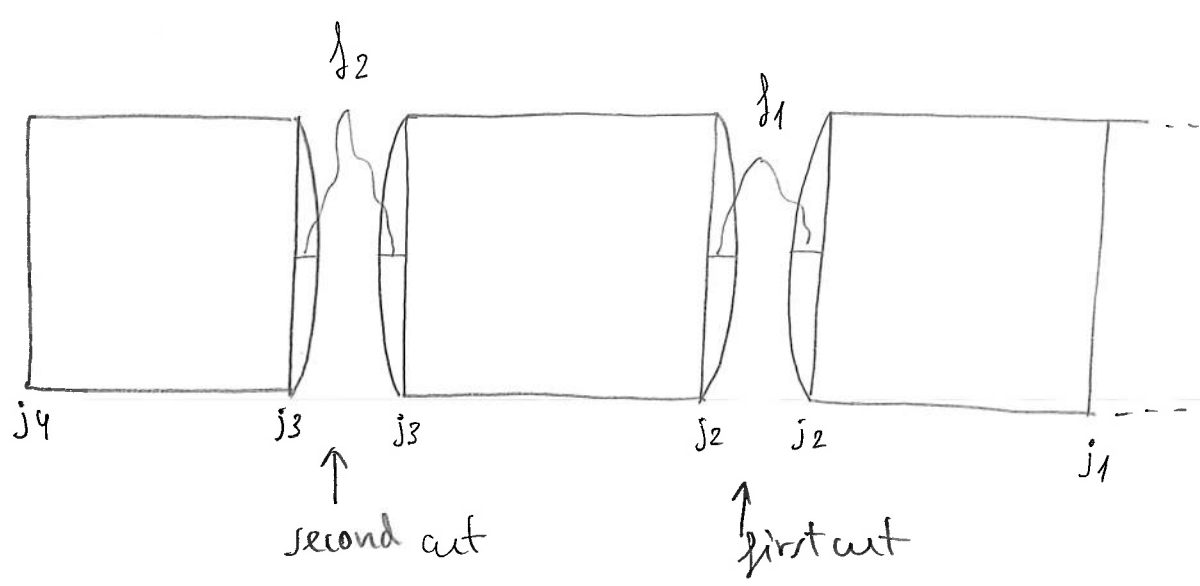
In all cuts, the left and right enlargements are the same

NOTE:

$2f = \Delta_4 \rightarrow f_3 = k_3 \frac{\Delta_4}{2}$

compatibility in cut intrados (see notes about compatibility in first and second cuts)

# Compatibility in first and second cuts



$$2f_1 + 2f_2 = \Delta_1 + \Delta_2 + \Delta_3$$

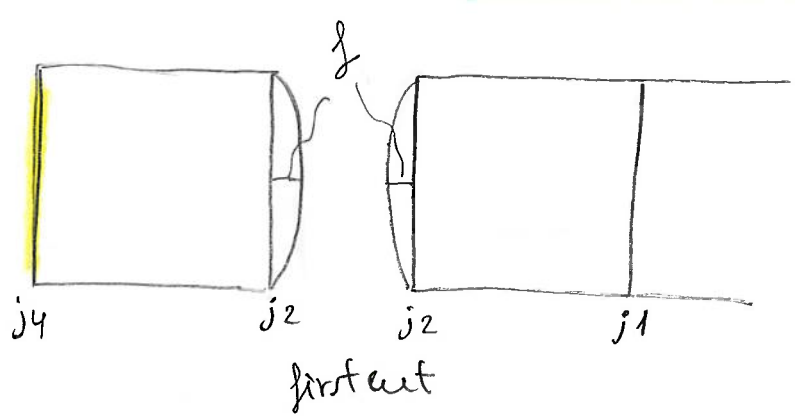
let's define :

$$f_1 = k_1 \frac{1}{4} (\Delta_1 + \Delta_2 + \Delta_3)$$

$$f_2 = k_2 \frac{1}{4} (\Delta_1 + \Delta_2 + \Delta_3)$$

coefficients  $k_1, k_2$  usually 1.0 but the designer may choose other values, such as  $k_1 = 0.0$   $k_2 = 0.0$  which eliminates the 3D-shaping. These coefficients control the depth of 3D-shaping!

## Case only one cut in extruder



$$2f = \Delta_1 + \Delta_3$$

$$f_1 = k_1 \left( \frac{\Delta_1 + \Delta_3}{2} \right)$$

EXAMPLE 1

EXAMPLE 3D SHAPING DATA FILE

```

*****
* 3D SHAPING
*****

```

```

1
1 → type "1"
upper 2 1
1 40 45 1.0
2 45 52 1.0
lower 1 1
1 60 65 1.0

```

extrados  
intrados

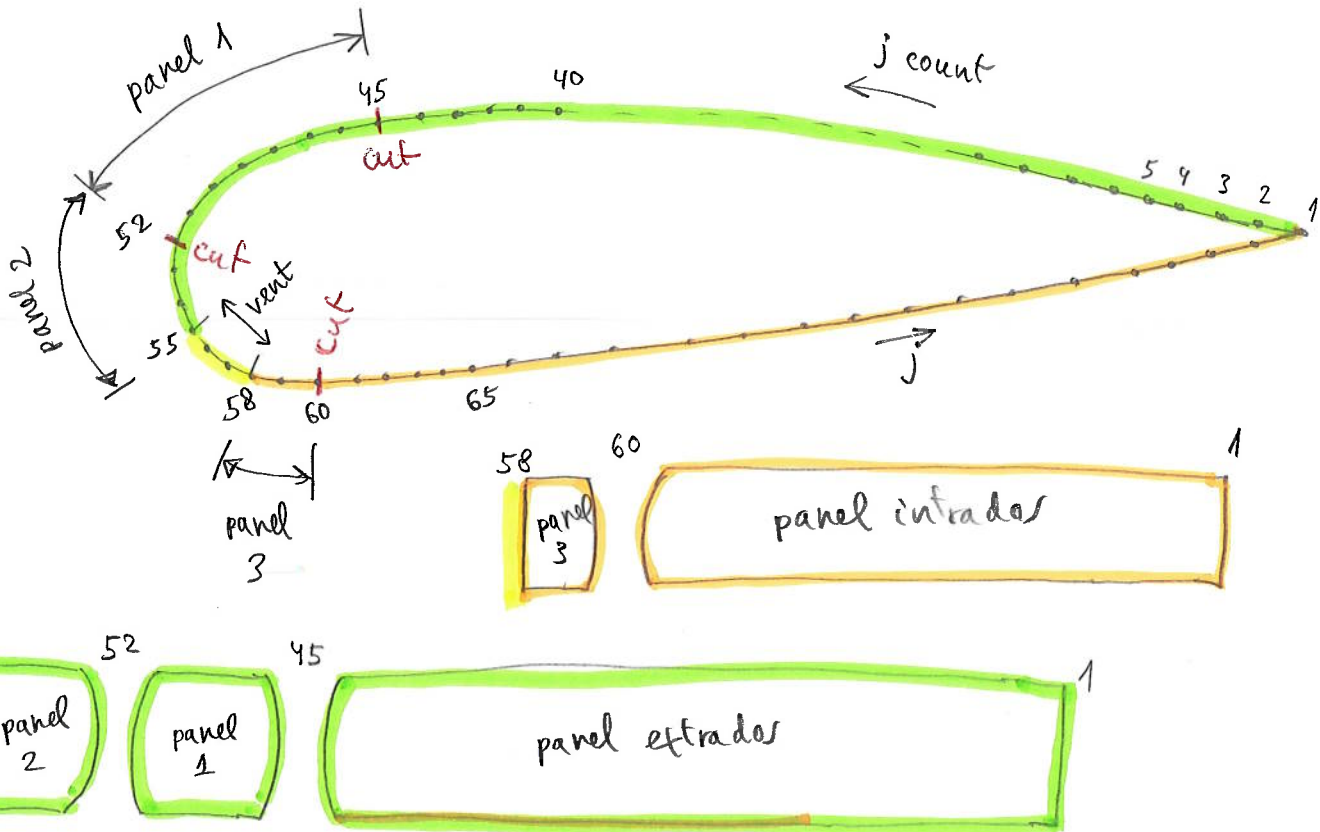
Yes, use 3D-shaping

Use two cuts type "1"

cut 1 in zone of influence 40 to 45, cut in 45, use k=1.0 depth of effect

cut 2 in zone 45 to 52, cut in 52 use 1.0 depth

cut 1 in point 60 and area extended to point 65 use one cut type "1" in lower surface





```

*****
* SECTION 29 3D SHAPING
*****
0

```

EXAMPLE 2

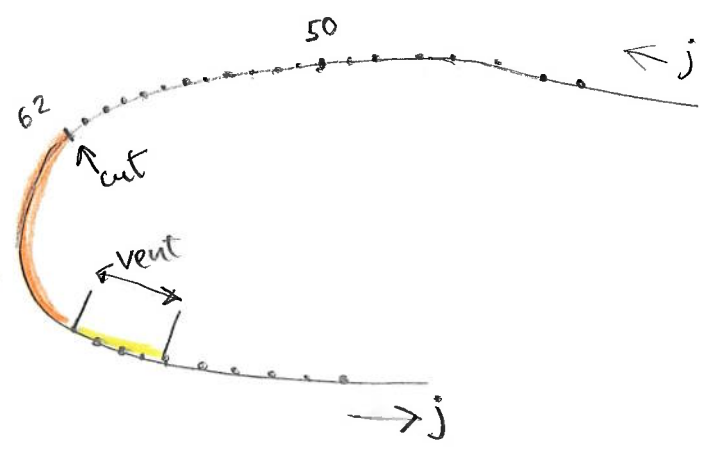
→ Do not use 3D-shaping

```

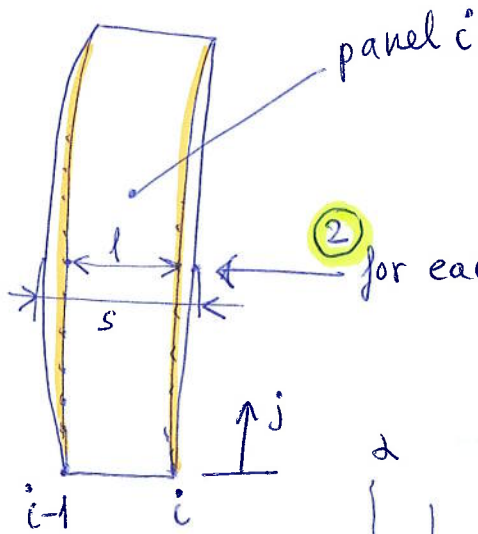
*****
* SECTION 29 3D SHAPING
*****
1 → Yes, use
1 → type "1" (only 1 available)
upper 1 1 → In upper surface use 1 cut type 1
1 50 62 1.0 → Cut Δ in z.i. from point 50 to 62, cut in point 62
lower 0 1 → In lower surface use 0 cuts

```

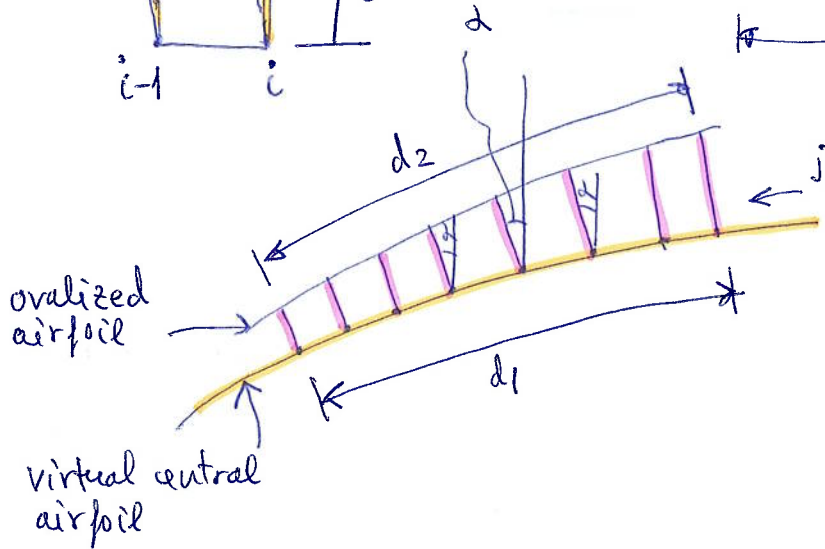
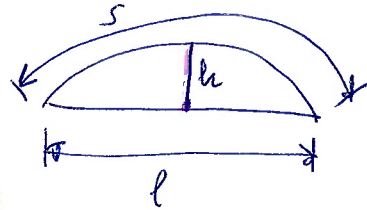
EXAMPLE 3



① For each rib  $i = 1, \dots, n_{max ribs}$



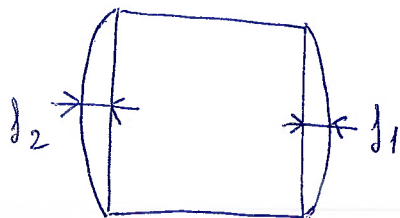
② for each  $j$  compute  $h, s, h, \alpha$



③ for each cut and zone of influence compute  $d_1, d_2, \Delta$   
 $\Delta = d_2 - d_1$

④ Assign border enlargements in each cut considering  $\Delta$  and  $k$  (depth of effect)

For each part



⑤ Separate and draw all parts  
 - panel contours  
 - sewing contours  
 - reference points  
 - special marks

⑥ Print summary report with the elongations applied to each cut

⑦ Generate file with tessellation .dxf and .stl (future implementation), to visualize 3D effects.